

Return to Flight Project Management Perspectives from the Tile Repair Project

March 31, 2004

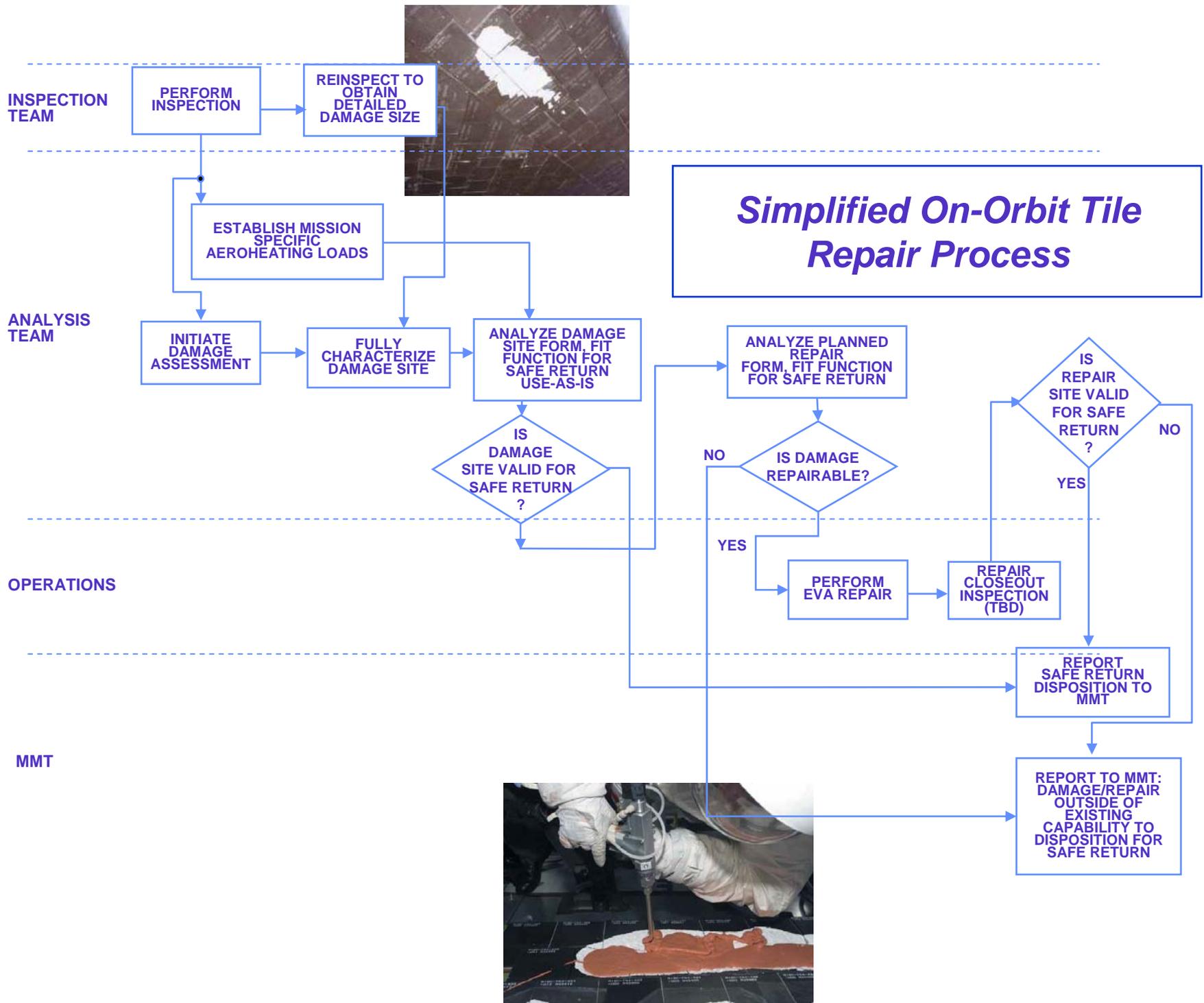


Tim Brady
Timothy.k.brady@nasa.gov
282-483-9118

- **Philosophy of the Return-to-Flight (RTF) has been to put in several layers of protection to prevent the reoccurrence of an accident**
 - *Eliminate/minimize sources of debris shedding*
 - *Understand tolerance of TPS systems to debris impact*
 - *Improve ground inspection capabilities*
 - *Develop on-orbit inspection capability*
 - *Develop on-orbit repair capability*
 - *Provide Contingency Shuttle Crew Support*

- We have to detect it
- We have to know if it is catastrophic or not
- We have to get access to the damage site to effect repair
- We have to apply a material with an EVA crewmember
- The material must be capable of withstanding the heat of reentry, and
- We have to have confidence, through sound engineering rationale, that the damaged or repaired vehicle can return with the crew safely

- **The tile repair project objectives for return to flight are to:**
 - *Develop and qualify material for use in on-orbit repair*
 - *Develop and certify EVA tools to deliver material to repair worksite*
 - *Develop analytical tools to provide real-time engineering support*
- **Ultimately repair capability is the ability to:**
 - *Provide technical rationale to support a use-as-is disposition (no repair required) for damaged tile*
 - *Provide physical on-orbit repair as required (EVA tools and techniques)*
 - *Provide technical rationale to support disposition of the repair, if required, for re-entry*



- **Material Selection**

- *Based historically on MA-25S material tested in 17979-1980 timeframe*
 - » *Development test focus: control swell, cured state fully porous, meet thermal performance*
- *RTV-based Cure-in-place ablator formed from two-part mixture (12 part resin: 1 part catalyst)*



Cured Material in Tile

- **Material Requirements**

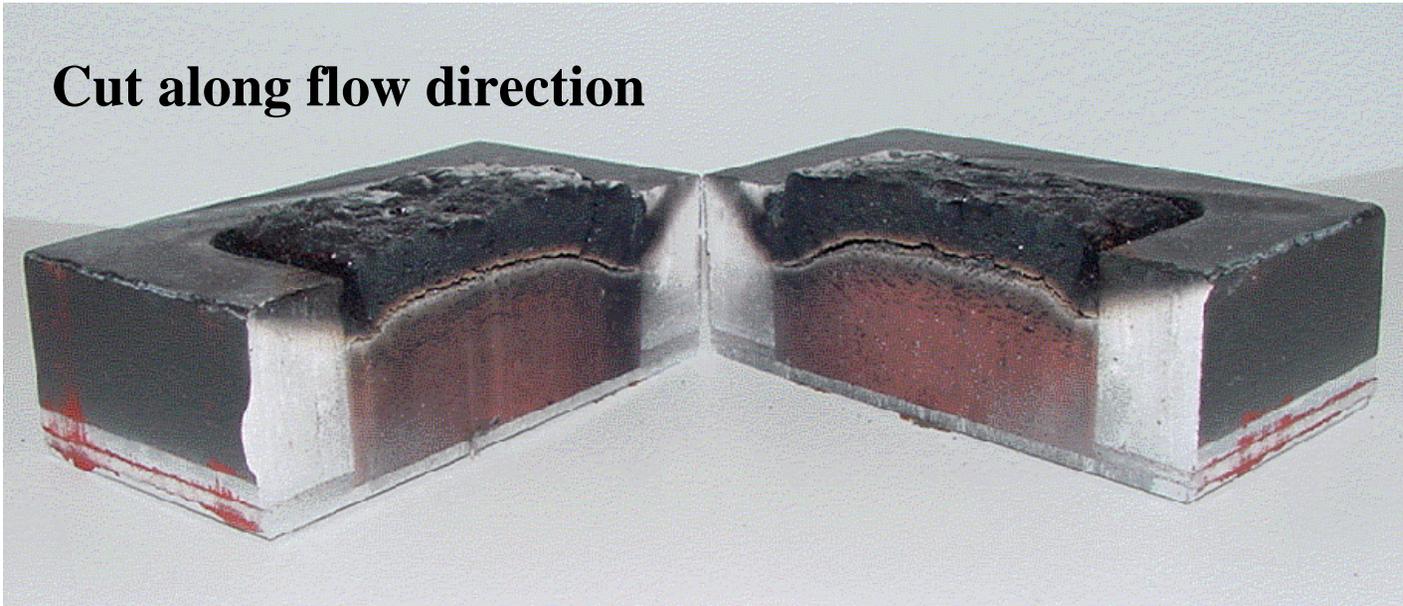
- *Fill an irregular volume*
- *Cure in a vacuum*
- *One hour working life*
- *Interface strength capable of withstanding reentry loads*
- *Survive thermal reentry environment*
- *Material swell within ± 0.25 inch of the outer mold line (OML) contour*
- *Provide thermal protection to aluminum substructure*



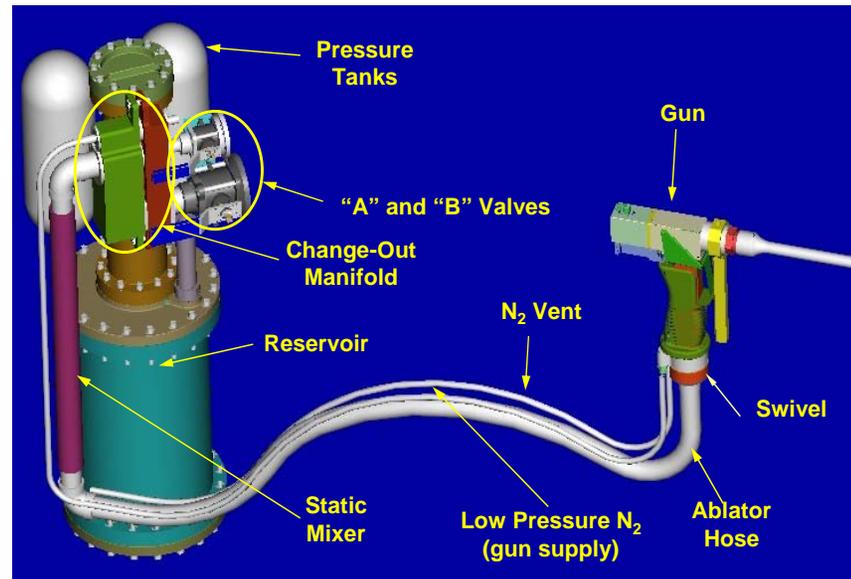
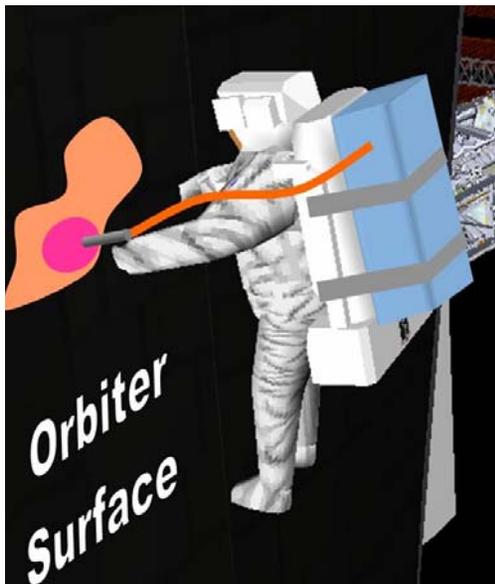
Post-Arc Jet Test

- Arc jet test simulates re-entry plasma environment
- Char layer forms on top surface
- Char layer swell protrusion inter the outer mold line contour must be minimized

Cut along flow direction



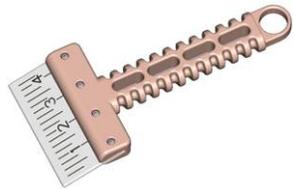
- **CIPA applicator to contain, mix, and extrude viscous, 2-part material**
 - *Vacuum-sealed material contained in two separated reservoirs*
 - *Pneumatic drive system employed to propagate materials through a static mixer and extrude the mix through a hose and gun/nozzle*
 - *EVA-removable/changeable mixer, hose, gun assembly*
 - *IVA and EVA heating and MLI provisions to maintain material temperature*
 - *Middeck volume packaging for launch/landing stowage*
- **Backpack to interface with the PLSS for on-orbit translation use**



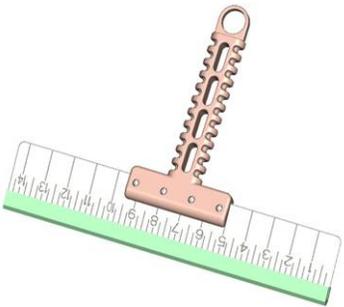
- Measurement/inspection tool for gauging pre- and post-repair depth and area dimensions
- Hardness measurement instrumentation
- Silicon brushes for tile-cavity cleaning prior to material injection
- Foam brushes to lightly smooth and remove voids in cavity
- Scrapers for removing excess material
- Flat and angled foam stamps for flattening, smoothing material
- Trash bags and EVA wipes for material clean-up and containment



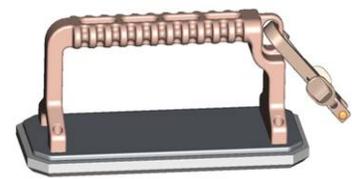
1.5" Trowel



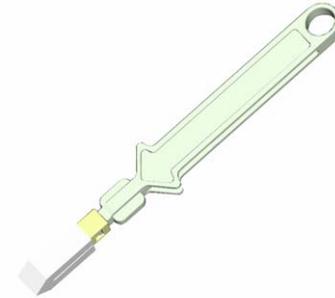
4.5" Trowel



14" Foam Trowel

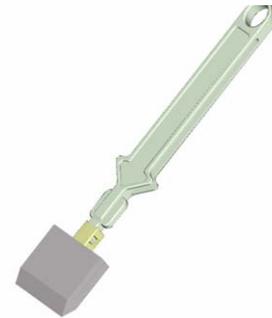


Flat Stamp

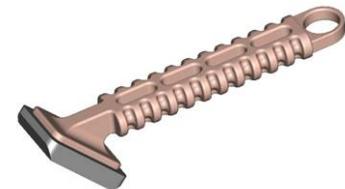


Foam Brushes

With Removable Brush Handle



Angled Stamp

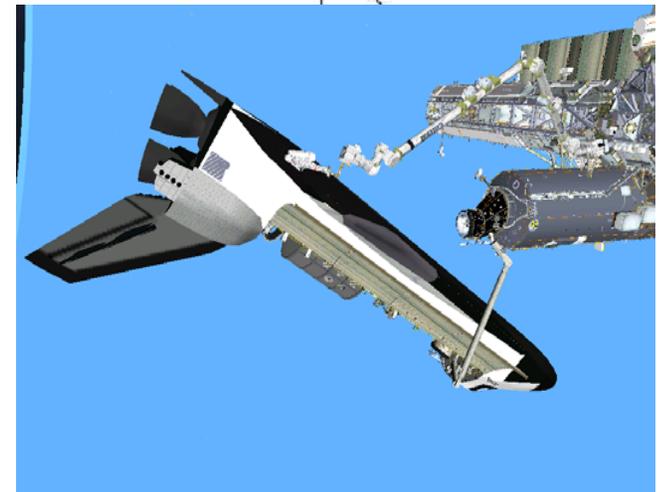
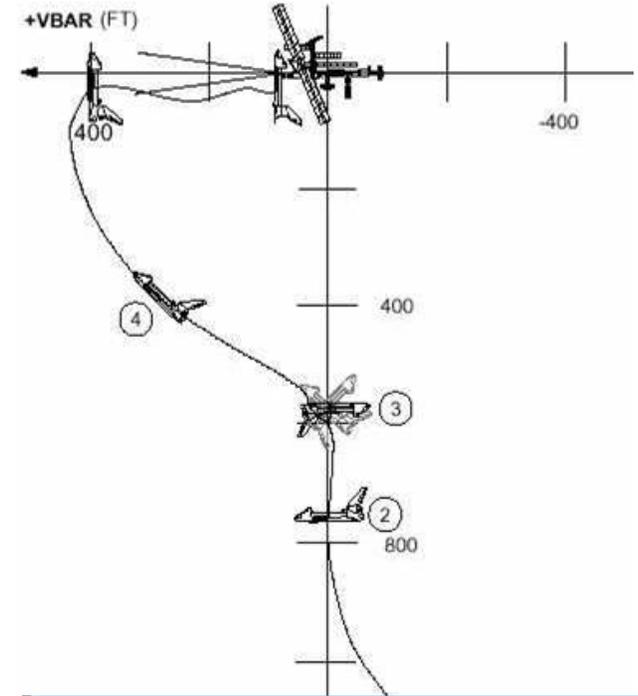


- **Inspection**

- *Tile is inspected from ISS (using cameras) with a pitch maneuver at 600 feet during approach*
- *A boom and sensor system attached on the end of the SRMS will be available to re-inspect damage areas to measure depth*
- *Data from these inspections used to determine if a repair is necessary and if a repair will be effective*

- **Access to repair site**

- *Grappled unberth from ISS using the Shuttle robot arm*
- *Orbiter repositioned for crewmember to access worksite while using the ISS robot arm*
- *Access available until JEM module launch*





**1. Trim Gap Filler as Required
Clean Tile with Gel Brushes**



2. Layer Material



3. Flatten / Smooth Repair

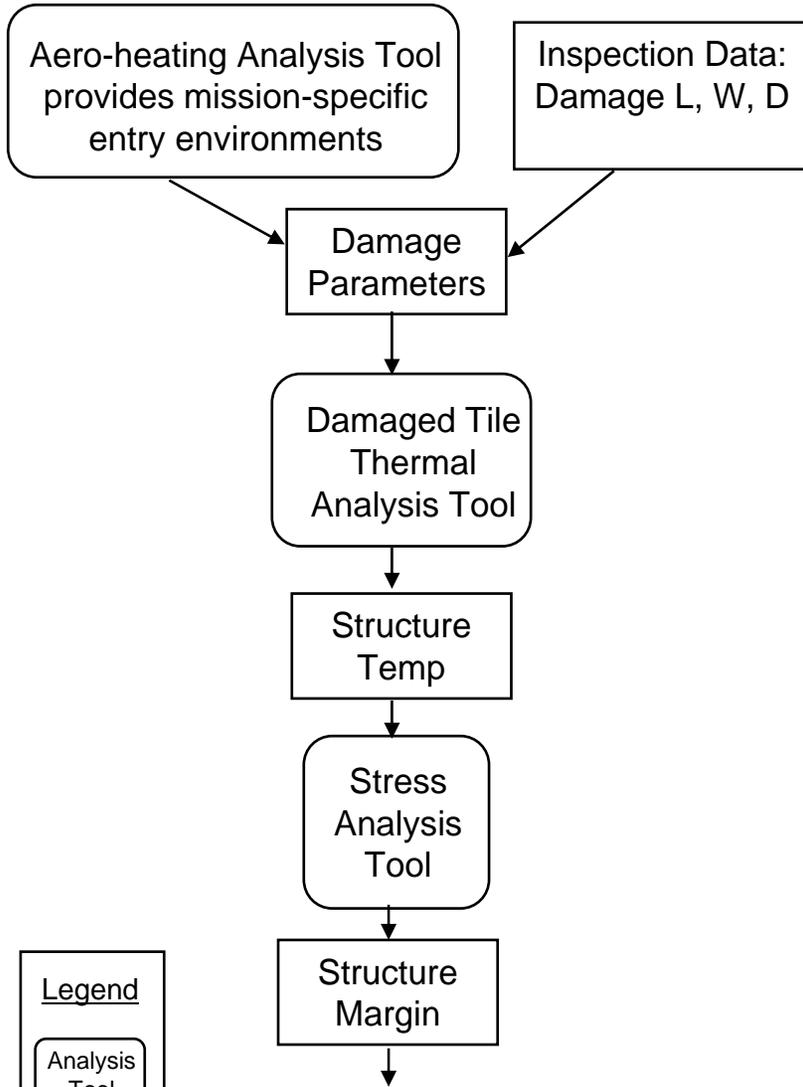


4. Ramp Trailing Edge



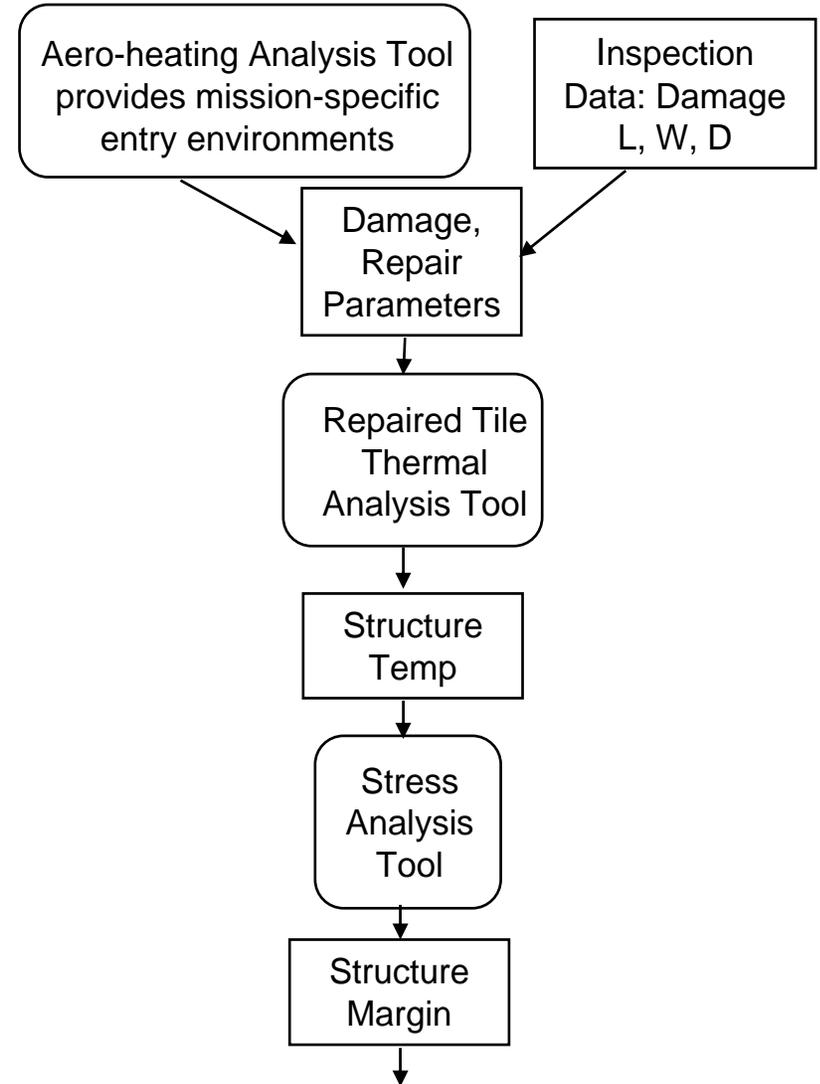
5. Verify Step

Damaged Tile Analysis

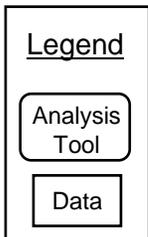


Repair/No Repair Decision

Repaired Tile Analysis



Within/Not Within Capability to Disposition for Safe Return



- **Prior to formally establishing tile repair as a project a flight techniques panel was established**
 - *Panel led by flight director Paul Hill*
 - *Major operational issues addressed*
 - *Technical feasibility testing and analysis initiated*
 - *Approximately 15 meetings over 6 months*

- **Flight techniques panel activities laid a solid foundation for the RTF activities**
 - *Springboard for boom sensor, tile repair and RCC repair projects*
 - *Technical development activities initiated*
 - *Identified major project interdependencies and technical risk*

Project Hurdles Remaining

Project Mid-term Lessons Learned

Backup Material

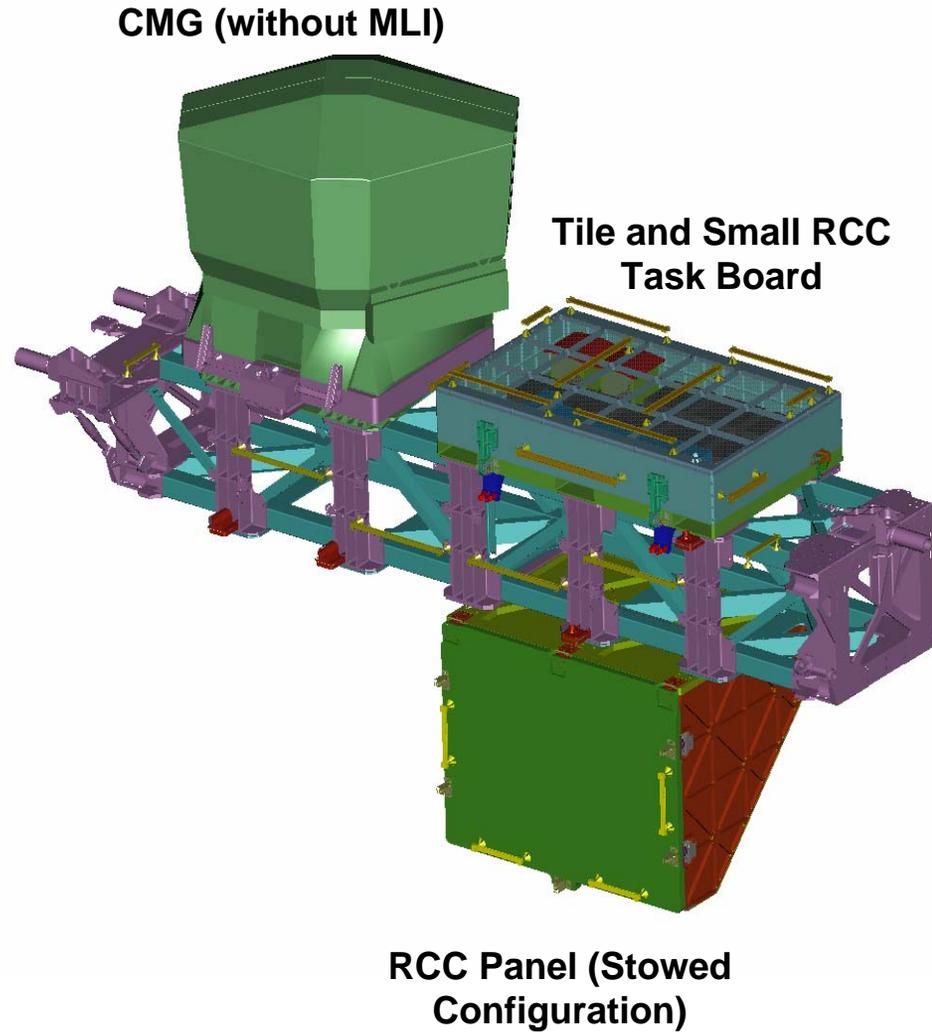
Tile and RCC Repair Flight Experiment

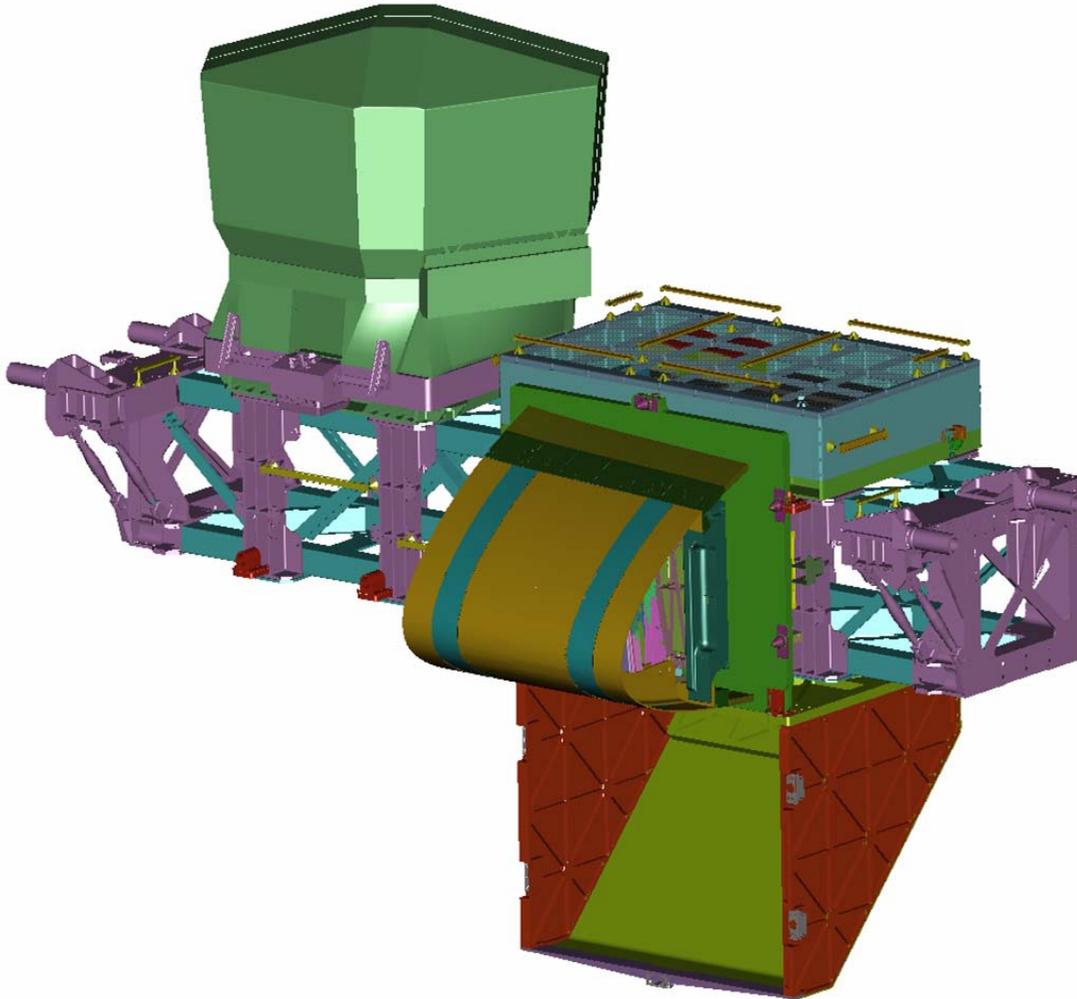
**Development Test Objective (DTO)
to be flown on STS-114 and STS-121**

- **Tile and RCC repair demonstrations are planned for the STS-114 and STS-121**
- **Tile repair objectives include:**
 - *To demonstrate various repairs (shallow, deep) using planned EVA tools and techniques*
 - *To demonstrate performance of an on-orbit repair through post-flight inspection, pull-test and/or arc jet testing*
 - *Investigate microgravity material performance; e.g. voids, cure, working life, foaming etc*
 - *Demonstrate repair of a large tile cavity (on the order of 10"x30")*
- **RCC objectives are dependent on final selection of repair concepts, however DTO hardware has been designed to provide capability to perform both small and large RCC repairs**
 - *Small repairs will be demonstrated with small sections of RCC*
 - *A full scale panel or simulated panel can demonstrate large scale repair techniques*

- **Top surface of the LMC:**
 - *Carrier for task board containing tile repair and small RCC repair test articles*

- **Lower surface of the LMC:**
 - *Large scale RCC repair: RCC panel, T-seals, carrier panel and attach structure mounted to the for launch and landing*
 - *Panel will be deployed for RCC repair operations*





- RCC Panel with T-seals and 6” extensions all the way around
- Hard mounted under the LMC for launch and landing
- Panel is hinged up and in front of the LMC for repair task